

NASA Human Spaceflight Conjunction Assessment: Recent Conjunctions of Interest

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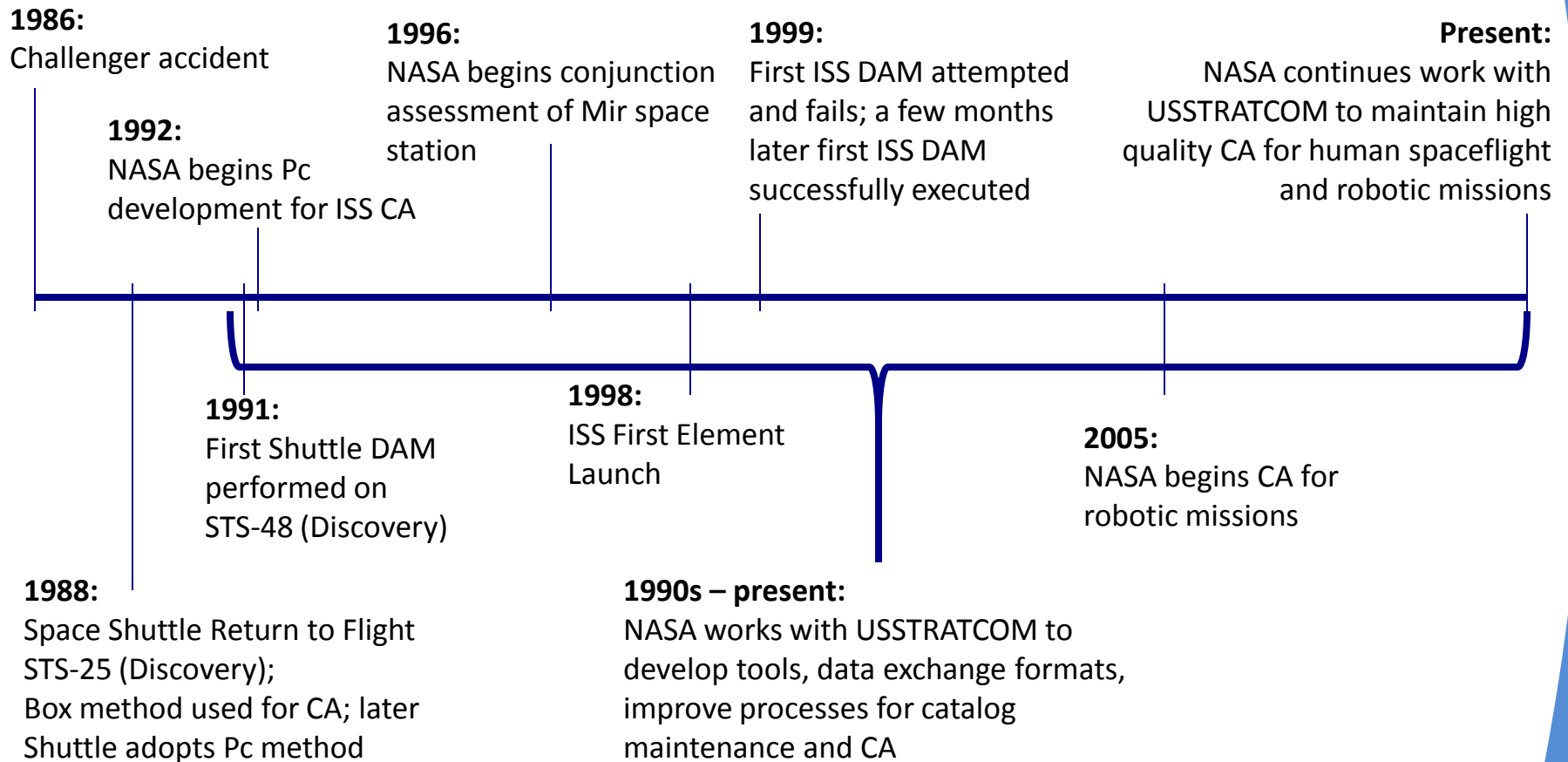
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Interface Working Group Chair

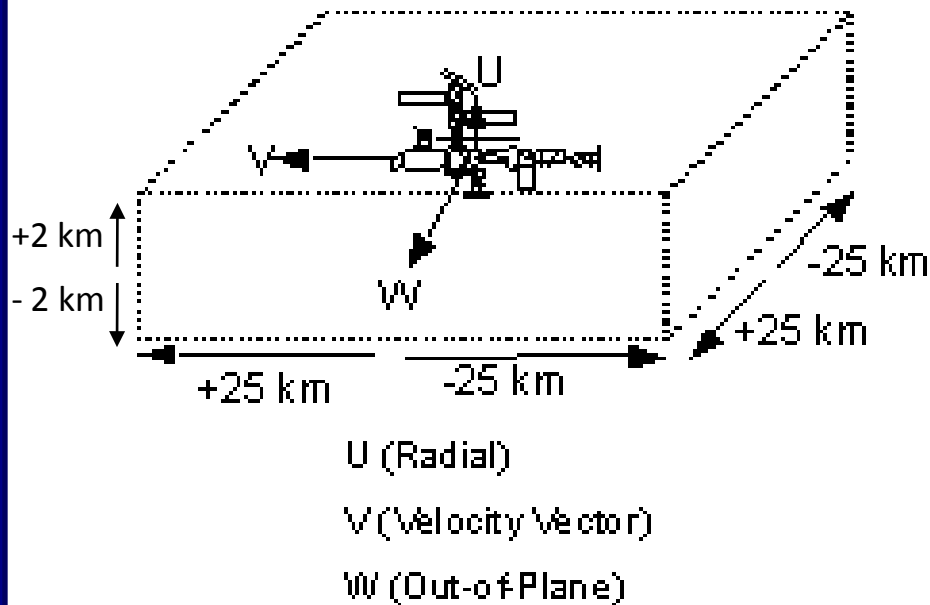
Presentation Overview

- Brief history of NASA Human Spaceflight Conjunction Assessment (CA) activities
- Overview of NASA CA process for ISS and Shuttle
- Recent examples from Human Spaceflight conjunctions

NASA Human Spaceflight Conjunction Assessment (CA) History



ISS and Shuttle Conjunction Screening and Notification



Screening Process

- Every 8 hours, JSpOC screens ISS/Shuttle against high accuracy catalog 72 hours into the future
- Depending on miss distances, JSpOC notifies NASA (see Screening Volumes below)
- NASA and JSpOC discuss each conjunction
- If object is a concern
 - JSpOC increases tasking on object
 - JSpOC provides NASA with more frequent updates

Screening Volumes (in Kilometers – U x V x W)

10 x 40 x 40: JSpOC automated notification – refine threat object solution – no NASA notification

2 x 25 x 25: JSpOC notifies NASA of conjunction – shown above

0.75 x 25 x 25: NASA notifies larger ISS team – only for ISS or joint ISS/Shuttle conjunctions

ISS and Shuttle Debris Avoidance Processes

- NASA/MCC-Houston (MCC-H) calculates Probability of Collision (P_c) upon notification and data from JSpOC
- MCC-H uses P_c along with additional conjunction data to make recommendation on Debris Avoidance Maneuver (DAM)
 - ISS
 - Decision is coordinated between ISS International Partners
 - Decision must be made no later than 24 hours prior to conjunction Time of Closest Approach
 - DAM performed with engines on the Russian Segment of ISS – no crew involvement
 - Shuttle
 - Decision can be made closer to Time of Closest Approach
 - Crew must be awake to perform DAM
- Flight Rules in place to determine when DAM should be performed based on P_c
 - Risk of conjunction is weighed against risk of mission safety and success
 - For the Shuttle only, a “box method” downmode is available to make a decision on DAM if P_c is not available

DAM Thresholds

ISS/Shuttle DAM Pc Thresholds

Pc

$1\text{E-}05 \leq$

Pc

$< 1\text{E-}04 \leq$

Pc

Shuttle-only DAM Box Method Downmode Thresholds

Yellow

Threshold

1 Km (U) x 7 Km (V) x 7 Km (W)

Red

Threshold

0.5 Km (U) x 4 Km (V) x 4 Km (W)

ISS Debris Avoidance Maneuver History

Date	Debris	Vehicle	Notes
06/13/99	SL-3 Rocket Body	FGB	Maneuver Execution FAILED. Two Red Pc violations early in event based on bad data
10/26/99	Pegasus Rocket Body	FGB	First successful ISS DAM and only DAM performed by FGB.
09/29/00	SL-3 Rocket Body	Progress	Yellow Pc threshold violation
02/10/01	Unknown Debris	STS-98	Shuttle Box method used
12/15/01	SL-8 Rocket Body	STS-108	Shuttle Pc method used
05/15/02	SL-8 Rocket Body	Progress	Red Pc threshold violation
05/30/03	MEGSAT	Progress	Red Pc threshold violation
08/27/08	COSMOS 2421 Debris	ATV-1	Red Pc threshold violation Largest PC calculated to date
03/22/09	CZ-4 Debris	STS-119	Red Pc threshold violations on consecutive orbits (~50) Conjunction orbits were during an EVA Retrograde DAM was executed early by having orbiter hold attitude
07/18/09	Unknown Debris	STS-127	Conjunction occurred 15 hours after STS-127 docking during crew sleep. Red Pc threshold violation post-docking DAM performed by the Shuttle before the crew went to sleep

More than 700 ISS conjunction notifications to NASA
 10 Debris Avoidance Maneuvers Attempted
 5 performed by ISS
 1 attempted by ISS, but failed (ISS first attempt)
 4 performed by Shuttle during mated operations

Annual Maneuver Rate (theory suggests ~1.2 maneuvers/year)
 ~0.9 Maneuvers/year (including failed DAM)
 ~0.8 Maneuvers/year (not including failed DAM)

Shuttle Debris Avoidance Maneuver History from 3 Shuttle Flights

Date	Debris	Flight	Notes
08/11/07	Delta Rocket Body	STS-118	No maneuver for very close conjunction Notification too late (18 minutes prior to time of conjunction)
03/11/08	USA-193 Debris	STS-123	Existing burn modified, used box method
03/25/08	Breeze-M Debris	STS-123	Existing burn modified, although no Pc violation
11/28/08	Cosmos 2421 Debris	STS-126	Separation burn delayed several hours to avoid Yellow Pc violation

- Typically Shuttle modifies existing burns for Debris Avoidance rather than adding a burn for a high risk conjunction
- On average, approximately 5-10 conjunction notifications are received during Shuttle mission
 - Of those, usually 1-2 at most are serious enough to require a modification to a burn

Recent ISS/Shuttle Conjunction Examples

August 27, 2008: COSMOS 2421 debris

- Notable for: First ISS DAM attempted in 5 years, Retrograde DAM
- Retrograde DAM – performing a posigrade DAM would violate Soyuz launch and landing constraints, as well as Shuttle rendezvous altitude constraints
- DAM Clearing Assessment was very difficult due to multiple conjunctions in the vicinity if the DAM was performed
 - Eventually found a 1.0 m/s option that was technically clear with respect to Flight Rules (no post-burn conjunction within 48 hours of a maneuver)
 - The DAM option chosen with the best prospects still had one object (different piece of Cosmos 2421 debris) that could be a potential problem 3 days into the future
- New COSMOS 2421 Debris
 - DAM was nominal for original debris which resulted in a predicted conjunction with another piece of Cosmos 2421 debris 3 days later
 - DAM planning to avoid the 2nd piece of debris began soon after the previous DAM executed
 - Ultimately, the Pc dropped below the yellow threshold and DAM was not performed

March 12, 2009: PAM-D debris

- Notable for: Late notification, crew placed in Soyuz
- Complicating factors
 - High eccentricity
 - High radial velocity
 - Extremely high drag
 - Low inclination
 - Small radar cross section
 - Space weather prediction of geomagnetic spike did not occur as expected
- Notification from JSpOC
 - TOPO notified at TCA-42 hours that a piece of PAM-D debris would enter 2x25x25 km box
 - Usual screening horizon is 72 hours
 - At TCA-19 hours, prediction entered 0.75 x 25 x 25 km box
 - At this point, TOPO informed Russians and ISS Flight Control Team
 - ISS DAM template kick off process NLT TCA-28.5 hours

March 23, 2009: CZ-4 debris

- Notable for: Retrograde mated DAM during STS-119
 - MCC-H had been monitoring this repeating conjunction as a “no threat” item
 - Object had similar orbital period to the ISS/STS stack, so there were TCAs on multiple consecutive orbits
 - ISS/STS stack had a Loss of Attitude Control at ~TCA-48 hours, which pushed the radial miss distances much closer (within the 0.75 x 25 x 25 km box)
 - Probability of Collision calculations showed multiple red threshold violations
 - TCAs would occur during the next spacewalk
 - The posigrade delta-V needed to be safe would violate the rendezvous conditions necessary for the upcoming Soyuz – a small retrograde maneuver was planned
 - DAM was unique in that it was performed using the Shuttle to hold attitude control such that an overall retrograde trajectory perturbation was accomplished
- Due to the retrograde DAM, this object ultimately showed up as a repeating conjunction again in the week following STS-119 undocking.
 - No PC threshold violations occurred and no action was necessary

April 9, 2009:

Fengyun 1C debris

- Notable for: DAM planning initiated on account of post-Soyuz undocking trajectory
- Complicating factors:
 - Pre-undock trajectory indicated no DAM planning would be required
 - Soyuz undocking at TCA-30 hours moved ISS to <1 km total miss from Fengyun at TCA (though larger radial miss)
 - Since collision probability requires JSpOC ISS covariance, MCC-H had to wait for several USSTRATCOM tracks
 - Immediate ISS GPS vector from indicated a likely low-Pc
 - DAM cancelled 4 hours later, as expected, following post-undock ISS tracking
- The object was relatively easy to track and predict

May 16, 2010:

Unknown Debris

- Notable for: TCA occurred ~1 hour after STS-132 docking
- NASA notified 72 hours prior to TCA
 - Approximately 24 hours prior to STS-132 launch
- Docking perturbations would invalidate the Pc method
- Not enough time post-docking to track the mated stack to get an accurate prediction
- Trajectory teams discussed the option of having ISS perform a stand-alone DAM the evening prior to docking. However, this would require ISS to make a decision to perform a DAM without official Pc.
- Trajectory teams discussed the possibility of having the Orbiter delay docking to ensure a good PC
- After NASA analysis, teams became comfortable that the perturbations from docking activities would not be large enough to statistically affect the overall risk

Questions?

